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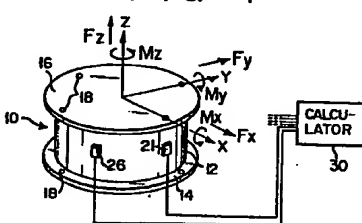
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(64) Sensor for sensing three orthogonal forces and three orthogonal moments.

(57) A sensor includes six strain gages (21, 22, 23, 24, 25, 26) attached to a detection section (12) equidistantly in the circumferential direction of the detection section. The strain gages are of an orthogonal two-gage type and, in an orthogonal coordinates system (X, Y, Z) with the axial direction of the detection section as the Z-axis, are arranged such that their strain detection direction is inclined at an angle  $\alpha$  of  $0 < \alpha < 45^\circ$  from the Z-axis. A linear relation given by the following matrix expression exists between outputs  $e_1, e_2, e_3, e_4, e_5, e_6$  of the six strain gages and six force components  $F_x, F_y, F_z, M_x, M_y, M_z$ :  
$$\begin{bmatrix} e_1 \\ e_2 \\ e_3 \\ e_4 \\ e_5 \\ e_6 \end{bmatrix} = [B] \cdot \begin{bmatrix} F_x \\ F_y \\ F_z \\ M_x \\ M_y \\ M_z \end{bmatrix}$$
 where [B] denotes a matrix of coefficient and can be initially found by applying known force components to the detection section and calibrating. It is therefore possible to calculate force components from this matrix expression.

**FIG. 1**



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Sensor for sensing three orthogonal  
forces and three orthogonal moments

This invention relates to a sensor for sensing three orthogonal forces and three orthogonal moments, which is attached to the wrist, etc., of robots to measure the three orthogonal forces and three orthogonal moments at that location and thus to control the load of the wrist, etc., with the use of these data items.

In the robots, it is necessary to provide a sensor on their wrist to measure forces and moments applying there and thus to control the load acting upon the wrist of the robot. In an orthogonal coordinates system (X, Y, Z) with the axis of the robot hand as the Z axis, the sensor measures forces  $F_X$ ,  $F_Y$  and  $F_Z$  of the X-, Y- and Z axes and moments  $M_X$ ,  $M_Y$  and  $M_Z$  around the X-, Y- and Z axes. The conventional sensor of this type is adapted to measure the respective forces and moments independently (See A.K. Bejczy, R.S. Dotson. Conf Proc. IEEE Southeastcon, 1982. pages 458 to 465 and G. Plank, G. Hirzinger, Int. Control Probl. Man. Tech. 1982 pages 97 to 102). In order to independently measure the respective forces and moments with as less a mutual interference as possible, eight strain gages are used so as to cancel the forces acting in a predetermined axial direction or to reduce them to a substantially negligible extent. Furthermore, the sensor structure is so designed as to permit a proper attachment of these

strain gages, thereto. That is, in the calculation of the respective axial forces and moments from values measured by the strain gages, a proper sensor structure is adopted to permit ready calculation of them with a lesser number of times and an ingenious design is also adopted to properly attach the gages to the portion of the robot. This arrangement involves a complex, bulkier sensor structure with no consequent practical advantage. Furthermore, a detection circuit cannot be incorporated in the sensor due to the complex structure and thus an extra space is required for the detection circuit to be provided outside the sensor. In order to produce a greater deformation, it is also necessary for the sensor to be made flexible in structure and thus the structure becomes complex in this respect.

It is accordingly the object of this invention to provide a sensor for sensing three orthogonal forces and three orthogonal moments, which is simple and compact in its structure and high in its practical application and assures a ready incorporation into robots or the like. This sensor is comprised of a detection section of a thin-walled cylindrical configuration which, when a load is applied to the sensor, generates strain; six strain gages arranged on the detection section at the corresponding six positions mutually separated in the circumferential direction of the detection section; and an arithmetic calculator for calculating six force components  $F_X$ ,  $F_Y$ ,  $F_Z$ ,  $M_X$ ,  $M_Y$ ,  $M_Z$ , exerting upon the detection section, from the six outputs  $\epsilon_1$ ,  $\epsilon_2$ ,  $\epsilon_3$ ,  $\epsilon_4$ ,  $\epsilon_5$ ,  $\epsilon_6$  of the six strain gages.

According to this invention, the six force components can be detected by attaching six strain gages to the detection section of the cylindrical configuration. This sensor is simple and compact in its structure, higher in its practical application and permits ready incorporation into a robot and the like. In this connection it is to be noted that the

conventional sensor is so properly designed as to separate the respective force components from the result of a strain detection. According to this invention, the detection values of six strain gages are obtained so as to find six force components (unknown forces), that is, as small a number of gages as necessary are used for that purpose, and the separate force components are obtained by arithmetically processing the detection outputs of the gages. Furthermore, the calculation process is relatively simple and it is not necessary to use an arithmetic calculator of any large capacity.

This invention can be more fully understood from the following detailed description when taken in conjunction with the accompanying drawings, in which:

Fig. 1 is a perspective view showing a sensor according to an embodiment of this invention;

Fig. 2 is a cross-sectional view showing the sensor shown in Fig. 1;

Fig. 3 is a cross-sectional view showing the sensor shown in Fig. 1;

Fig. 4 is a fragmentary, perspective view in cross section showing the sensor shown in Fig. 1;

Fig. 5 is a diagram for explaining a stress acting upon a strain gage; and

Fig. 6 is a view showing the direction in which the strain gage is sensitive.

In Figs. 1, 2 and 3 are shown a sensor 10 according to a first embodiment of this invention. The sensor 10 is comprised of a strain detection section 12 of a thin-walled cylindrical configuration and a flange 14 provided on one longitudinal end of the detection section 12 and extending radially and outwardly of the detection section 12. An end plate 16 of a disc-like configuration is attached to the other end of the detection section 12 and has a diameter larger than that of the detection section 12. Bolt poles 18 for mounting are formed on the flange 14 and on the extending portion

of the end plate 16. Through the bolt holes 18, the sensor 10 is bolted to a portion, such as the wrist of a robot, for which its load is detected. A deformation occurs at the strain detection section 12 according to the loads being applied to the flange 14 and end plate 16. The flange 14 and end plate 16 are thickly formed so as to provide a rigidity. On the other hand, the strain detection section 12 is thinly formed to allow the occurrence of a greater strain and thus to improve the SN (signal to noise) ratio.

Around the outer periphery of the strain detection section 12, six strain gages 20 (21, 22, 23, 24, 25, 26) are equidistantly attached to the detection section 12 along the outer periphery of the detection section 12. Thus, an angle made by the adjacent two strain gages with respect to the center of the detection section 12 is  $60^\circ$ . The respective strain gages detect strains at the locations of their attachments independently of each other. The outputs of the strain gages are supplied to the arithmetic calculator 30 where force components applied to the sensor 10 are calculated in the following way.

Here, an orthogonal coordinates system (X, Y, Z) is adopted where the Z axis represents a central axis of the detection section 12; the X axis, a direction located perpendicular to the Z axis and passing through the deformation gage 21; and the Y axis, a direction perpendicular to the Z and X directions. From these force components (forces and moments) being applied to the sensor 10 can be expressed by forces  $F_X$ ,  $F_Y$ ,  $F_Z$  acting in the corresponding directions and moments  $M_X$ ,  $M_Y$ ,  $M_Z$  acting around the respective axial directions.

As shown in Fig. 4, in a plane perpendicular to the Z axis, a minute element 4 is located in a direction made at an angle  $\theta$  with respect to the X axis. Stresses acting at two points on the minute element 4 are expressed by a normal stress  $\sigma_Z$  in the Z axis direction

and shear stress  $\tau_{\theta Z}$  as shown in Fig. 5. Since there is no pressure difference between the inside and outside of the detection section 12, no circumferential stress acts upon the minute element 4 and thus  $\sigma_{\theta}$  is normally 0.

5 The axial stress  $\sigma_Z$  is expressed, as the function of the angle  $\theta$ , by Equation (1) below:

$$\sigma_Z(\theta) = F_Z/S + \{(M_X - F_Y L)\sin\theta - (M_Y - F_X L)\cos\theta\}/Z$$

... (1)

where

S and Z: the lateral, cross-sectional area and section modulus, of the cylindrical section of the detection section 12; and

L: the distance between the end plate 16 and the minute element 4.

On the other hand, the shear stress  $\tau_{\theta Z}$  is expressed, as the function of the angle  $\theta$  representing the position of the minute element 4, as follows:

$$\tau_{\theta Z}(\theta) = M_Z/IS + (F_Y \cos^2 \theta - F_X \sin^2 \theta)/\pi r t$$

(0  $\leq$   $\theta$   $\leq$   $\pi/2$ ) ... (2)

$$\tau_{\theta Z}(\theta) = M_Z/IS - (F_Y \cos^2 \theta + F_X \sin^2 \theta)/\pi r t$$

(25  $\pi/2 \leq \theta \leq \pi$ ) ... (3)

$$\tau_{\theta Z}(\theta) = M_Z/IS - (F_Y \cos^2 \theta - F_X \sin^2 \theta)/\pi r t$$

(30  $\pi \leq \theta \leq 3\pi/2$ ) ... (4)

$$\tau_{\theta Z}(\theta) = M_Z/IS + (F_Y \cos^2 \theta + F_X \sin^2 \theta)/\pi r t$$

(35  $3\pi/2 \leq \theta \leq 2\pi$ ) ... (5)

In this way,  $\sigma_Z$  is the function of  $F_X$ ,  $F_Y$ ,  $F_Z$ ,  $M_X$  and  $M_Y$  and  $\tau_{\theta Z}$  is the function of  $F_X$ ,  $F_Y$  and  $M_Z$ . If, therefore, only  $\tau_{\theta Z}$  of three  $\theta$ 's (for example,  $\theta_1$ ,  $\theta_3$ ,  $\theta_5$ ) are detected, then three ( $F_X$ ,  $F_Y$  and  $M_Z$ ) of six force components are found from Equations (2) to (5). If only  $\sigma_Z$  of the remaining three  $\theta$ 's ( $\theta_2$ ,  $\theta_4$ ,  $\theta_6$ ) is

detected, it is possible to find the remaining  $F_X$ ,  $M_X$   
and  $M_Y$  from Equation (1), since  $F_X$ ,  $F_Y$  and  $M_Z$  are  
already known. In this case, in order to find only the  
normal stress  $\sigma_Z$  in the Z axis direction, the strain  
gages 22 ( $\theta_2$ ), 24( $\theta_4$ ) and 26( $\theta_6$ ) are so attached to the  
detection section 12 that their strain-sensitive  
direction is the Z direction. Then, the normal stress  
 $\sigma_Z$  is found from Equation (6) below.

$$\sigma_Z(\theta) = E \cdot \epsilon_Z(\theta) \quad \dots(6)$$

On the other hand, in order to find only the shear  
stress  $\tau_{\theta Z}$ , the strain gages 21( $\theta_1$ ), 23( $\theta_3$ ) and 25( $\theta_5$ )  
are so attached to the detection section 12 that their  
strain-sensitive direction is 45°-inclined with respect  
to the Z-axis direction. In this case, it is only  
necessary to find, from Equation (7) below, the normal  
strain which is produced in the 45° direction.

$$\tau_{\theta Z}(\theta) = 2G \cdot \epsilon_T(\theta) \quad \dots(7)$$

where E and G stand for the young's modulus and shear  
modulus, respectively.

Where only the shear stress is to be found, merely  
attaching the strain gage to the detection section 12 at  
an angle 45°-inclined to the axial direction will result  
in the axial strain component of Equation (1) being  
contained in the strain detection quantity. Therefore,  
a two-gage method using two gages in an orthogonal array  
is employed to detect strain. By so doing, the axial  
strain component is cancelled and it is therefore  
possible to detect strain only in the 45°-inclined  
direction.

If stress (and thus strain) is to be detected using  
such orthogonal gages, the strain components, which vary  
depending upon the temperature, cancel each other and it  
is possible to obtain a detection value free from strain

components resulting from the temperature variation. It is therefore preferable that, in order to obtain a temperature-compensated measured value, not only a strain in the 45°-inclined direction but also a strain  
5 in the Z-axis direction be measured using the orthogonal gages. Where use is made of the orthogonal gages, an amount of strain in the 45°-inclined direction can be detected as double the strain amount and an amount of strain (axial, perpendicular stress) in the Z-axis  
10 direction can be detected as  $(1 + \nu)$  times the strain amount, noting that  $\nu$  is the poisson's ratio.

In order to further improve the detection sensitivity, it is only necessary to attach the respective orthogonal gages to the inner surface and  
15 outer peripheral surface of the cylindrical detection section 12. When this is done, an amount of strain is amplified to four times the strain amount in the 45°-inclined direction and to  $2(1 + \nu)$  times the strain amount in the Z-axis direction. Actually, it is  
20 considered that there are manufacturing and strain gage attaching errors. It is, therefore, preferable to, through the application of known loads or moment to the sensor, find a conversion equation from a relation of the known loads or moment to the actual measured value  
25 and evaluate a measured value from the conversion equation.

According to the sensor thus constructed, forces and moments ( $F_X$ ,  $F_Y$ ,  $F_Z$ ,  $M_X$ ,  $M_Y$  and  $M_Z$ ) applied at the respective locations of the sensor 10 can be readily  
30 evaluated through a simple calculation with the use of six strain gages attached to the sensor. This sensor is simpler in its structure and it is possible to selectively determine the attaching position, dimension, etc. of the sensor 10. It is also possible to dispose  
35 the detection circuit, etc. on the inner surface of the cylindrical detection section 12. This sensor 10 finds a wider application.



Another embodiment of this invention will be described below.

In the first embodiment, the three of the six strain gages are so arranged that their strain detection direction is the Z-axis direction with the detection direction of the other three strain gages 45°-inclined with respect to the Z axis. This arrangement permits the separate detection of the X-direction normal stress and shear stress. In the second embodiment, on the other hand, gages in an orthogonal array are used as strain gages 21, 22, 23, 24, 25 and 26 and, as shown in Fig. 6, the detection direction of the respective strain gages is inclined from the Z-axis by an angle  $\alpha$  ( $0 < \alpha < 45^\circ$ ). In the second embodiment, the detection value of the respective strain gages contains all the strains resulting from the six force components ( $F_X$ ,  $F_Y$ ,  $F_Z$ ,  $M_X$ ,  $M_Y$  and  $M_Z$ ). In other words, in this embodiment, the strain resulting from the six force components is intentionally contained in the detection value by inclining the detection direction of the respective gages by the angle  $\alpha$ .

The respective two outputs of the six strain gages 21, 22, 23, 24, 25 and 26 so attached to the sensor 10 are expressed by Equations (8) and (9) below:

$$\begin{aligned} \epsilon a(\theta) = & \epsilon_Z(\theta) (\cos\alpha - v\sin\alpha) \\ & + \epsilon_T(\theta) \{\cos(\pi/4-\alpha) - \sin(\pi/4-\alpha)\} + \epsilon\Delta T \\ & \dots(8) \end{aligned}$$

$$\begin{aligned} \epsilon b(\theta) = & \epsilon_Z(\theta) (-v\cos\alpha + \sin\alpha) \\ & + \epsilon_T(\theta) \{\sin(\pi/4-\alpha) - \cos(\pi/4-\alpha)\} + \epsilon\Delta T \\ & \dots(9) \end{aligned}$$

As the output of the respective gages,  $\epsilon(\theta) = \epsilon a(\theta) - \epsilon b(\theta)$  is produced, noting that, as evident from Equations (8) and (9), the strain component  $\epsilon\Delta T$  resulting from a variation of temperature is not contained in  $\epsilon(\theta)$ . With the angle  $\alpha$  set to be 0,

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$\epsilon(0) = (1+\nu)\epsilon_z$ . With the angle  $\alpha$  set to be  $45^\circ$ ,  $\epsilon(45) = 2\epsilon_T$ . The axial normal stress  $\sigma_z$  and shear stress  $\tau_{\theta z}$  can be found from Equations (6) and (7) by detecting the strain  $\epsilon(0)$  in the Z-axis direction and strain  $\epsilon(45)$  in the  $45^\circ$ -inclined direction. This is a principal on which the force components are calculated in the first embodiment.

With the second embodiment, on the other hand, since the angle  $\alpha$  is not  $0^\circ$  and  $45^\circ$ , both  $\epsilon_z$  and  $\tau_{\theta z}$  are contained in  $\epsilon(\theta) = \epsilon a(\theta) - \epsilon b(\theta)$ . Thus, the respective gages, since being six in number, can produce the following outputs:

$$\begin{aligned} \epsilon(\theta_1) &= \epsilon_1; \epsilon(\theta_2) = \epsilon_2; \epsilon(\theta_3) = \epsilon_3 \\ \epsilon(\theta_4) &= \epsilon_4; \epsilon(\theta_5) = \epsilon_5; \epsilon(\theta_6) = \epsilon_6. \end{aligned}$$

Substituting Equations (6) and (7) into Equations (8) and (9), six relational expressions are obtained between the outputs of the strain gages and the respective six unknown quantities  $\sigma_z(\theta)$  and  $\tau_{\theta z}(\theta)$  [ $\theta = \theta_1 \dots \theta_6$ ]. The relations as shown in Equations (1) to (5) exist between the force components  $F_X, F_Y, F_Z, M_X, M_Y, M_Z$  and the  $\sigma_z(\theta)$  and  $\tau_{\theta z}(\theta)$  [ $\theta = \theta_1 \dots \theta_6$ ]. In consequence, a linear relation given by Equation (10) below is obtained between the outputs of the six strain gages and the sixth force components:

$$\begin{aligned} (\epsilon_1, \epsilon_2, \epsilon_3, \epsilon_4, \epsilon_5, \epsilon_6)^T &= [B] \cdot (F_X, F_Y, F_Z, M_X, \\ &M_Y, M_Z)^T \dots (10) \end{aligned}$$

where

T: the transposed vector

[B]: the (6 x 6) matrix of coefficients

By applying an individual force component to the sensor, the relation between the force components and the outputs of the strain gages is found for the respective force component and the respective element can be

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evaluated from the relation through a calibration.

According to the second embodiment, six force components  $F_X$ ,  $F_Y$ ,  $F_Z$ ,  $M_X$ ,  $M_Y$  and  $M_Z$  can be found through a simple matrix calculation all at one time from the outputs  $\epsilon_1$ ,  $\epsilon_2$ ,  $\epsilon_3$ ,  $\epsilon_4$ ,  $\epsilon_5$ ,  $\epsilon_6$  of the six strain gages which are equidistantly attached in six places to the outer periphery of the cylindrical detection section 12. In the second embodiment, it is not necessary to adopt any particular design, such as adjusting the strain-sensitive direction of the strain gages, so as to individually detect the normal stress in the Z-axis direction and shear stress. For this reason, the sensor structure becomes more simplified and finds a wider range of practical application.

This invention is not restricted to the above-mentioned embodiments. The method for detecting deformations at six places on the detection section is not restricted to the above-mentioned two-gage method. It is also possible to use another proper method, such as a four-gage method. In the above-mentioned embodiments, the gages are equidistantly located at six places on the outer periphery of the detection section. This is because a high independence of each strain gage is assured with respect to the other strain gages. If, however, the respective strain gages are arranged on the outer periphery of the detection section, it is still possible to obtain the six force components. Furthermore, the singular point of the strain detection value can be avoided by arranging adjacent strain gages to the detection section in the longitudinal direction of the detection section. In this way, a variety of changes and modifications can be made without departing from the spirit and scope of this invention.

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Claims:

1. A sensor for sensing three orthogonal forces and three orthogonal moments, comprising:  
a detection section (12) which produces strain when a load is applied to the detection section; and  
5 strain gages (21, 22, 23, 24, 25, 26) arranged at the detection section;  
characterized in that  
the detection section is of a thin walled cylindrical shape,  
10 the strain gages are six and spaced in the circumferential direction of the detection section, and  
an arithmetic device (30) calculates, from outputs  $\epsilon_1, \epsilon_2, \epsilon_3, \epsilon_4, \epsilon_5$  and  $\epsilon_6$  of the six strain gages, six force components  $F_X, F_Y, F_Z, M_X, M_Y$  and  $M_Z$  which are  
15 applied to the detection section.
2. A sensor according to claim 1, characterized in that, in an orthogonal coordinates system (X, Y, Z) where the Z-axis is in the axial direction of said detection section and X- and Y-axes are perpendicular to  
20 each other and to the Z-axis, force components  $F_X, F_Y, F_Z, M_X, M_Y$  and  $M_Z$  are such that said force components  $F_X, F_Y$  and  $F_Z$  are forces acting in the X-, Y- and Z-axis directions and that the force components  $M_X, M_Y$  and  $M_Z$  are moments acting around the X-, Y- and Z-axis moments,  
25 respectively.
3. A sensor according to claim 2, characterized in that said six strain gages (21, 22, 23, 24, 25, 26) are arranged equidistantly on said detection section (12).
4. A sensor according to claim 2, characterized in  
30 that three strain gages (21, 23, 25) are of an orthogonal two-gage type, each of which have their strain direction  $45^\circ$ -inclined with respect to the Z-axis and the remaining strain gages (22, 24, 26) are arranged in a direction parallel to the Z-axis; and only a shear  
35 stress  $\tau_{\theta Z}$  is found from the outputs  $\epsilon_1, \epsilon_3$  and  $\epsilon_5$  of

5           5. A sensor according to claim 4, characterized in  
that said arithmetic calculator (30) calculates six  
force components from equations (1) to (5) below:

$$\sigma_z(\theta) = F_z/S + \{(M_x - F_y L)\sin\theta - (M_y - F_x L)\cos\theta\}/Z \dots(1)$$

$$\tau_{\theta Z}(\theta) = M_Z/rS + (F_Y \cos^2 \theta - F_X \sin^2 \theta)/\pi r t \quad (0 \leq \theta \leq \pi/2) \quad \dots(2)$$

$$\tau_{\theta Z}(\theta) = M_Z/rS - (F_Y \cos^2 \theta + F_X \sin^2 \theta)/\pi r t \quad (\pi/2 \leq \theta \leq \pi) \quad \dots (3)$$

$$\tau_{\theta Z}(\theta) = M_Z/rS - (F_Y \cos^2 \theta - F_X \sin^2 \theta)/\pi r t \quad (\pi \leq \theta \leq 3\pi/2) \quad \dots(4)$$

$$\tau_{\theta Z}(\theta) = M_Z/rS + (F_Y \cos^2 \theta + F_X \sin^2 \theta)/\pi r t \quad (3\pi/2 \leq \theta \leq 2\pi) \quad \dots(5)$$

S and Z: the lateral cross-sectional area and section modulus of the cylindrical portion of said detection section; and

L: a constant showing the position of the strain gage.

6. A sensor according to claim 5, characterized in that said arithmetic calculator (30) calculates three force components  $F_X$ ,  $F_Y$  and  $M_Z$  from three measured values for the shear stress  $\tau_{\theta Z}$  on the basis of said equations (2) to (5) and calculates the remaining force components  $F_X$ ,  $M_X$  and  $M_Y$  from three measured values for the normal stress  $\sigma_r$  on the basis of the equation (1).

7. A sensor according to claim 2, characterized in that each of the strain gages is of an orthogonal

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two-gage type.

8. A sensor according to claim 2, characterized in that the respective strain gages (21, 22, 23, 24, 25, 26) are of a four-gage type and are arranged two on the inner periphery surface and two on the outer periphery surface of said detection section (12).

9. A sensor according to claim 2, characterized in that the respective strain gages (21, 22, 23, 24, 25, 26) are of an orthogonal two-gage type and are arranged on the detection section (12) such that their strain detection direction is inclined at an angle  $\alpha$  of  $0 < \alpha < 45^\circ$  with respect to the Z-axis.

10. A sensor according to claim 9, characterized in that the output  $\epsilon(\theta)$  of the respective strain gage (21, 22, 23, 24, 25, 26) is obtained as a difference between two outputs  $\epsilon a(\theta)$  and  $\epsilon b(\theta)$  and the outputs  $\epsilon_1, \epsilon_2, \epsilon_3, \epsilon_4, \epsilon_5$  and  $\epsilon_6$  of the six strain gages ( $\theta = \theta_1, \theta_2, \theta_3, \theta_4, \theta_5, \theta_6$ ) are found through the calculation of the following matrix equation:

20  $(\epsilon_1, \epsilon_2, \epsilon_3, \epsilon_4, \epsilon_5, \epsilon_6)^T = [B] \cdot (F_X, F_Y, F_Z, M_X, M_Y, M_Z)^T$   
where

[B]: a (6 x 6) matrix of coefficient

T: a transposed vector.

11. A sensor according to claim 10, characterized in that said matrix of coefficient is found based on the output of the respective gages (21, 22, 23, 24, 25, 26) through the application of known force components to the detection section (12).

FIG. 1

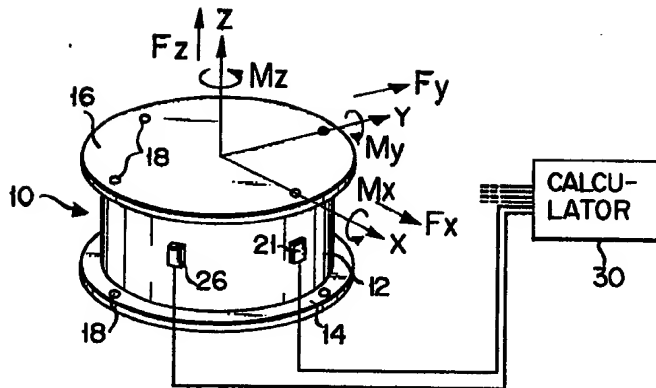


FIG. 2

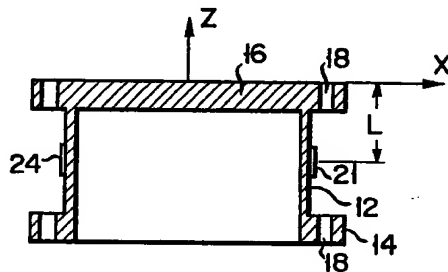


FIG. 3

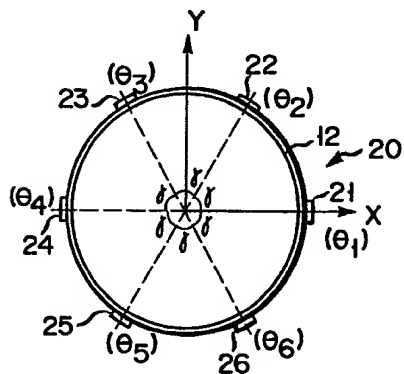


FIG. 4

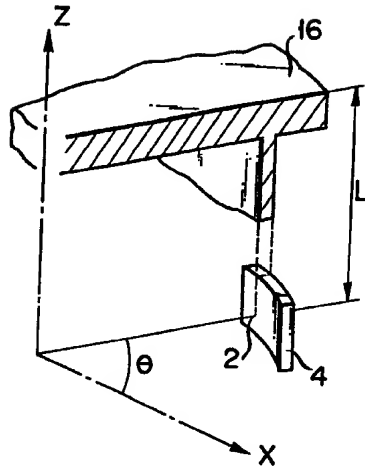


FIG. 5

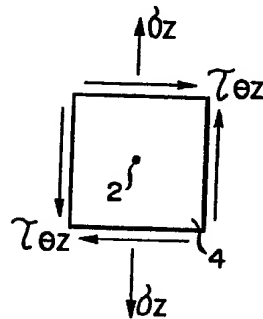


FIG. 6

